



COMPUTER MODELING AND OPTIMIZATION OF OBOGS WITH CONTAMINANTS

ANNUAL REPORT Covering the Period 15 July 1984 - 14 July 1985

Joseph J. Beaman

September 13, 1985

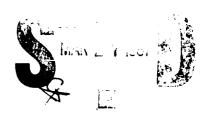
Supported by

U.S. ARMY MEDICAL RESEARCH AND DEVELOPMENT COMMAND

Fort Detrick, Frederick, Maryland 21701-5012

Contract No. DAMD17-84-C-4076

The University of Texas at Austin Department of Mechanical Engineering Austin, Texas 78712



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REPORT DOCUMENTATION P					Form Applied OMB No 1004-0188 Exp. Date: Jun 30, 1986
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FOREWORD

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SUMMARY

This report covers the first year of a two year project dealing with the development of an OBOGS computer model to aid in design and specification of these oxygen generation systems for the Army's in-flight, medivac, and field hospital use. A model which predicts the performance of a two bed system for 02 - N2 separation has been delivered and installed on the U.S. Army Aeromedical Research Laboratory (USAARL) VAX computer. A model to predict the propagation of contaminants in these beds is currently under development.

STATEMENT OF PROBLEM

The ultimate goal of this research is to insure proper design of molecular sieve oxygen generation systems for the U.S. Army's in-flight, medivac, and field hospital use. Specifically the research involves further development of an OBOGS model to include the effects of contaminants in the feed air. This OBOGS model can be used to optimize and design OBOGS systems with respect to system parameters such as cycle time and bed and valve dimensions.

BACKGROUND

The elimination of the need for liquid oxygen storage for in-flight and portable ground use by means of oxygen generation equipment is an especially attractive goal to the military. Such an accomplishment would not only increase safety, but would relieve the logistics requirements of LOX supply which is costly and can restrict mission durations. Although other means are available, the molecular sieve method of generating enriched oxygen breathing gas is the most feasible for in-flight usage. The molecular sieve system weighs little, and has the potential of high reliability.

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A molecular sieve system utilizes the process of physical adsorption to separate air into its components. Although a detailed discussion of adsorptive phenomena is beyond the scope of this report, a brief overview of the process is in order.

Adsorption is the process by which particular types of molecules contained in a liquid or gas are selectively collected onto a solid surface. The term "adsorbent" refers to the collecting material while the "adsorbate" is that material which is collected. In physical adsorption (as opposed to chemical adsorption), Vander Waal's forces create the bonds between adsorbent and adsorbate; thus, the interaction is weak. The solid adsorbent is typically a highly porous, granular material with interior pores of diameters on the order of .4nm, and with the total pore volume approaching 50% of that of the whole particle. Clearly, abundant internal storage space is available in the sieve material.

Adsorptive separation is the process by which the molecular sieve system would separate air into its components and thereby produce an enriched oxygen product gas.

To maintain continuous operation from an adsorption plant, molecular sieve beds must be periodically "regenerated" by a period of desorption in which the adsorbate is removed from the sieve material. There are several ways in which

desorption may be accomplished. "Thermal Swing" involves heating the bed to a temperature at which the adsorptive capacity of the bed material is so low that the adsorbate must leave the molecular sieve internal surfaces. "Pressure swing" creates the same effect by reducing the adsorptive capacity at essentially constant temperature. "Inert purge stripping" removes the adsorbate at constant temperature and pressure by the passage of fluid containing no adsorbate molecules, in which the adsorbate is soluble. "Displacement desorption" occurs when a fluid containing a high concentration of a more strongly adsorbed molecule is passed through the bed. The pressure swing adsorptive cycle, PSA, is of primary concern in this work.

Two important technological developments of recent decades have created the possibility of in-flight and portable air separation systems. Firstly, the domestic development and production of synthetic zeolites, a molecular sieve material, affords the availability of reproducible adsorbents. Secondly, the inception and development of the pressure swing adsorption cycle (PSA) has been found to yield a satisfactory production system.

The PSA process usually employs two or more molecular sieve beds and cycles them in a manner such that a continuous flow of enriched oxygen product gas is attained. Sufficient gas is produced to meet demand and to regenerate exhausted beds via pressure swing method. By use of an appropriate zeolite, oxygen or nitrogen (for fuel tank inerting) can be concentrated to almost any value up to 95% for 02and 99.8% for N2. The product gas can be delivered at normal room temperature and at some nominal pressure as desired. The process cost is basically the cost of air compression for the feed gas.

Traditionally, PSA systems have found applications in industry and have been very large systems which were relatively insensitive to the various rate phenomena of the adsorption process. In the development of a scaled-down system for portable applications, knowledge of transient behavior becomes exceedingly important due to the vastly increased sensitivity of system output to rate phenomena and the requirements of minimal size and weight. Additionally, the effects of potentially harmful contaminants and chemical agents which might be encountered during usage must be better understood.

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APPROACH TO THE PROBLEM

CONTAMINANT MODEL

As a first step in the modeling of an OBOGS system, an experimentally verified model of a single bed system containing a zeolite molecular sieve with contaminants in the inlet air is being developed. A model that has been previously developed for two component 02 - N2 gas mixtures is described in Preliminary work with 02 - N2 - C0 and 02 - N2 - C02appendix A. mixtures indicates that a similar model with a third component might be effective for contaminant studies. USAARL is presently conducting single bed experiments with contaminants such as CO2, CO, C2H6O (ethanol), NH3 (ammonia). C5H12 (pentane), CH3Br (methyl bromide), NO2, N204, 03 (ozone), CH4 (methane), CCl2F2 (freon 12), acrolein, oil breakdown products, and other battlefield products. These experiments are essentially column breakthrough experiments which will be used to validate the computer model.

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Once the dynamics of a single bed are correctly modeled, an overall system model for a two bed system with valves and mixing plenum will be developed. This modeling has been done for 02 - N2 mixtures. A description is contained in appendix B. A similar model is being developed for contaminant studies.

RESULTS AND CONCLUSIONS

A computer program of the OBOGS model described in appendix B has been delivered and installed on the VAX* computer at USAARL. This program is menu driven for ease of operator use and has REGIS* graphical output for use with VT125* or VT240* graphical terminals. The operator has a choice of output of on-line plotting of oxygen mole fraction or inlet/outlet mass flowrate versus time or off-line plotting of a dynamic simulation of OBOGS for one cycle. The operator can also select and change a variety of geometric and material parameters in order to aid in design studies. Sample inputs and outputs of the program are given in appendix C.

A preliminary 3 component OBOGS has been developed for contaminant studies. Before any further development on this program can continue the results of the experimental single bed data generated by USAARL is necessary. As soon as this data is available, it will be incorporated into a single bed model. Then a two bed OBOGS system model will be developed and experimentally verified.

Tradename of Digital Equipment Corporation

Single Bed Model Technical Description
This article is reprinted from the TRANSACTIONS OF THE ASME, JOURNAL OF DYNAMIC SYSTEMS, MEASUREMENT, AND CONTROL. The authors are J.J. Beaman, A.J. Healey, and J. Werlin, Dec. 1983

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A Dynamic Model of a Molecular Sieve Bed With Nonlinear and Coupled Isotherms¹

A mathematical model of the O_2 - N_2 separation process in a molecular sieve bed was developed and experimentally verified. Oxygen and nitrogen breakthrough experiments were performed on a single bed packed with molecular sieve 5A. In these experiments, the bed had initially adsorbed either pure oxygen or nitrogen and then a constant inlet flow of nitrogen or oxygen was switched on the bed. The outlet oxygen mole fraction and mass flowrate were then recorded. This was done at several pressures and flowrates. The model predictions agreed with the experimental data when a linear oxygen isotherm and a nonlinear Langmuir isotherm for nitrogen were used and when these were coupled. The numerical scheme presented in this study to solve the system equations makes efficient use of computation time.

I Introduction

The elimination of the need for liquid oxygen for crew breathing requirements in aircraft is an especially attractive goal. A means of accomplishing this goal is now available through the use of a molecular sieve, OBOGS (On-Board Oxygen Generation System). Molecular sieves are in widespread use industrially [1], but the special problems associated with an OBOGS have led to the need for a reliable computer-based simulation tool that would model the relatively rapid transient response of such systems.

One promising system uses a synthetic zeolite to adsorb nitrogen from engine bleed air, thereby producing oxygen enriched breathing gas. This system uses a "pressure swing" adsorption cycle [2] in order to achieve gas separation. In this cycle, nitrogen is adsorbed from engine bleed air at high pressure (producing oxygen enrichment) and then desorbed and exhausted to the armosphere at low pressure in order to replenish the sieve material. While large scale industrial systems are in commercial use, size and weight optimized units for airborne applications are still in the development stage.

In order to better understand the operation and scaling laws of the OBOGS, a computer model of this system has been developed. As a first step in this development, a model of a single molecular sieve bed with identified parameters is needed. This paper addresses the formulation and verification of such a model for predicting the transient response of gas

flow rate and gas concentration at the bed outlet to sudden changes in gas composition at the bed inlet.

Previous work in this area has treated the diffusion process to be instantaneous [3, 4] or has considered one component at the trace level [5]. This work considers a complete "washout" of one gas type using another. In other words, mole fractions at the inlet undergo step changes from 0 to 1 and vice versa, while the bed back-pressure and inlet flows are essentially constant.

II Theoretical Model

Figure 1 shows a schematic of the experimental single bed which was used to validate the molecular sieve bed model, where component A is oxygen, B is nitrogen, x is oxygen mole

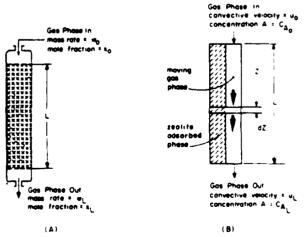


Fig. 1 Single molecular sieve bed

This work supported by Brooks Air Force Base, Aerospace Medicine, Crew Technologs

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Contributed by the Dynamic Systems and Control Division and presented at the Winter Annual Meeting, Washington, D.C., November 14-19, 1981, of The AMERICAN SHORTY OF MECHANICAL ENGINEERS. Manuscript received by the Dynamic System, and Control Division, February 3, 1982.

Journal of Dynamic Systems, Measurement, and Control

DECEMBER 1983, Vol. 105 / 265

sieve bed is 25.4 cm \times 14.0 cm \times 14.5 cm, loaded with 3.8 kg of Linde 5A sorbent screened through a 20 to 40 meshes/in. This corresponds to zeolite particles 340 to 833 microns in diameter. The void fraction of the bed is approximately $\epsilon = 0.36$. All experiments were run at room temperature of 25°C.

On the supply side of the bed were cylinders of pressurized On and No. Inflow was regulated by standard reducing valves and two variable orifice needle valves, each of which controlled the flowrate out of one cylinder only. A three-way tap valve was used to produce the step change in gas composition. Pressure transducers were located at the bed inlet and outlet. These transducers were located at the bed inlet and outlet. These transducers served as a check on the isobaric assumption. The amount of pressure drop across the bed was on the order of 3 percent of inlet pressure. A mass flowmeter was located just beyond the bed outlet to measure outflow. Outlet 0 mole fraction was measured with a respiratory mass spectrometer. Experimental data were taken for various flowrates at four different outlet pressures: 1/2 atm, 1 atm, 2 atm, and 3 atm. A hypobaric chamber was employed for the 1/2 atm test, while a hyperbaric chamber served as the plenum chamber for the 2 and 3 atm tests. For more information on the tests see Werlin [17].

Figure 3 shows the general nature of the oxygen and nitrogen "breakthrough" experiments where the symbols refer to experimental data. From 0 to 16 seconds an oxygen breakthrough is shown. As seen in this figure, the outlet mass flowrate rises to approximately twice the inlet flowrate and slowly trails off to its equilibrium value. The outlet mole fraction concentration starts at pure nitrogen and rises slowly to pure oxygen. At 16 seconds, the inlet flow is suddenly switched to pure nitrogen. As this point the outlet flow drops almost immediately, is constant for a while, and, finally it rises back to its equilibrium value as the concentration "front" propagates through the bed. The concentration time profile is much sharper than the oxygen breakthrough case, and at approximately 24 seconds the oxygen has been completely washed out.

This general behavior can be explained by the fact that 5A sorbent has a larger affinity for nitrogen than oxygen, so, when oxygen is washed out, more nitrogen is adsorbed than oxygen is given up. Therefore, the flowrate will be reduced ahead of the concentration front. The concentration front remains sharp due to the fact that the velocity ahead of the front is lower than that behind it.

For oxygen breakthrough, the effect is reversed. In this case, more nitrogen is desorbed than oxygen is adsorbed. There is a higher flowrate ahead of the front, and oxygen and nitrogen will tend to be convected away from this region, leading to a broader concentration front.

IV Identification of Model Parameters and Comparison With Experimental Data

Effect of Model Parameters. A quantitative assessment of the effect of the model parameters on the characteristic flowrate and mole fraction time plots can be made by writing (5) through (8) in nondimensional form as

$$\frac{\partial \omega}{\partial r} + 1 = g \tag{24}$$

$$\frac{\partial(\omega x)}{\partial t} + x = g_4 \tag{25}$$

$$\frac{\partial \eta}{\partial \tau} + g = 1 \tag{26}$$

$$\frac{\partial \eta_A}{\partial x} + g_A = x \tag{27}$$

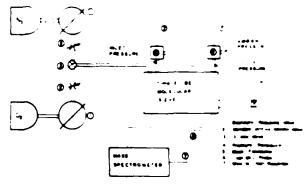
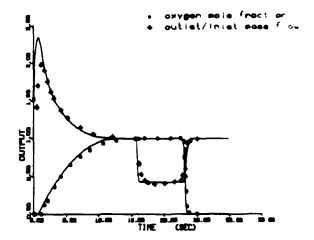


Fig. 2 Experimental apparatus



INLET FLOW (GM/SEC) = 4 28 PRESSURE (HPA) = .0507 KA = .160 KB = .059 d = 90.

Fig. 3 $\frac{1}{2}$ Atmosphere breakthrough data and simulation – linear isotherms

where

 $\omega = u/u_0$, $u_0 =$ inlet convective velocity, $\zeta = \beta Z/u_0$, $\eta = n/n_0$, $n_i =$ initial adsorbed phase concentration, $\eta_A = n_A/n_i$, $\tau = tK_Ad$, $x = C_A/C$, $g = \gamma/C$, and $g_A = \gamma_A/C$.

The single component isotherms of (18) and (19) can be expressed as

$$g_A^0 = n_A^0 \tag{28a}$$

$$g^0 = (1-\mu)\eta_B^0/(1-\alpha\eta_A^0)$$
 (28b)

where $\mu = 1 - K_B/K_A$ and $\alpha = n_0/b$. The ideal adsorption method is then used to couple these isotherms. For a nitrogen breakthrough, the initial and boundary conditions become

$$\omega(0,\tau) = 1$$

 $x(0,\tau) = 0$
 $\eta(\zeta,0) = 1$
 $\eta_A(\zeta,0) = 1$

Equations (24) through (28) with these conditions can then be solved numerically, as discussed previously, for values of the two parameters μ and α . These simulations indicate that the nondimensional distance down the bed is related to nondimensional travel time by

$$\zeta = (1 - \mu + \alpha)(\tau_{50} - 0.70) \tag{29}$$

where τ_{50} is the nondimensional time for the outlet oxygen

Table 1 Nitrogen breakthrough experimental results

P Bed pressure MP ₄	H' ₀ Inlet flowrate g∴s	H' _d Mass flow droop H' _{min} /H' ₀	time to 50% s
0.0507	3.21	0.41	9.40
0.0507	4.28	0.43	7.35
0.100	4.28	0.50	10.7
0.100	8.40	0.52	5.80
0.204	6.53	0.63	15.0
0.203	8.46	0.65	12.2
0.303	6.42	0.70	19.5
0.303	8.56	0.72	14.9

mole fraction to drop to 0.5 or 50 percent. This linear relation arises due to the fact that the outlet velocity drops to a value

$$\omega_d = 1 - \mu + \alpha \text{ for } 0 < \mu - \alpha < 1, \tag{30}$$

where ω_d is the droop velocity as determined by simulation and seen in Fig. 3 for t > 16 seconds. This velocity remains relatively constant for a time determined by the length of the bed. (The 0.70 in the (30) above comes from an intrinsic exponential time constant in the process, i.e., $e^{-.70} = 50$ percent.) For a given value of τ_{50} , the length at which the mole fraction "front" is 50 percent would be approximately: length = droop velocity x time to 50 percent which is the interpretation of equation (29).

Identification of Model Parameters. The values of K_A , K_B , b, and d can be effectively found by using breakthrough data and the results of the preceding section. The steps to obtain these parameters from a nitrogen breakthrough are:

- (1) Obtain values of $\mu \alpha$ from flowrate droop data, i.e., $\mu \alpha = 1 \omega_d$ for at least two different bed pressures. This allows calculation of μ and α .
- (2) Use (29) to obtain a value of K_A from the time for 50 percent breakthrough. The value of d is superfluous to the calculation since $\beta = d(1 \epsilon)/\epsilon$ and $\tau_{50} >> 0.70$ for the system under study. Using these two relations, (29) yields

$$K_A = L(1 - \epsilon)/u_0(1 - \mu + \alpha)\epsilon t_{50}$$

- (3) Use $K_B = (1 \mu)K_A$ to calculate K_B .
- (4) From dimensional analysis, one sees that x is a function of ζ and τ and therefore

$$\frac{\partial x}{\partial t} = \frac{\partial x}{\partial \tau} \frac{\partial \tau}{\partial t} = K_4 d \frac{\partial x}{\partial \tau}$$

This implies that the slope or tangent of x(L, t) is directly proportional to d. This fact can then be used to adjust d to match the mole fraction slope.

This resulting identification procedure can be checked by comparing oxygen breakthrough data and additional nitrogen breakthrough data to simulation predictions.

In Table 1, the experimental results of nitrogen breakthrough experiments of $\frac{1}{2}$, 1, 2, and 3 atmospheres are presented. Since more than the minimum two different pressures are available (theoretically all droop data should be identical at the same pressure), a least squares fit of all the droop data was used to find values of μ and α . Note $\omega_d = W_d M_{N_1} M_0$, where $M_{N_2} = 28$ for nitrogen and $M_{0_2} = 32$ for oxygen. A value of K_4 is then found from step 2 as

$$K_4 = L(1-\epsilon)\rho_0 A_4 / W_0 (1-\mu+\alpha)t_{50}$$

where ρ_0 is the bed inlet gas density. From step 3, $K_B = (1 - \mu)K_A$. Using slope adjustment as indicated in step 4, the mass transfer coefficient d was set at a value of 90 s⁻¹ at 1/2 atm and $W_0 = 4.28$ gm/s. This value is adjusted for pressure and flow rate by using a correlation due to Lightfoot [18] which indicates that d is proportional to W^{49}/P where W is mass flowrate and P is pressure. In any case, the value of d did not

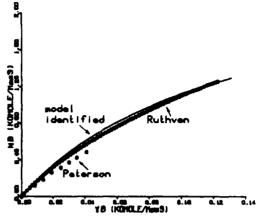


Fig. 4 Nitrogen isotherm

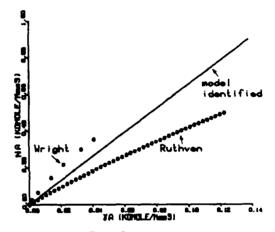


Fig. 5 Oxygen isotherm

have a large effect on the characteristic response. The result of this identification procedure gave the following parameter identification:

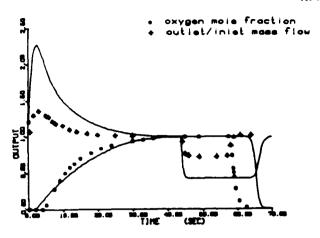
$$K_A = 0.153$$
 $\sigma_{KA} = 0.0185$ $K_B = 0.0506$ $\sigma_{AB} = 0.00611$ $\sigma_{AB} = 0.00611$

The o's are the predicted standard deviations of the parameters.

As a validation of this procedure for identifying isotherm parameters a comparison with published isotherms and correlations is shown in Figs. 4 and 5. Plotted in Fig. 4 are the nitrogen isotherm data of Peterson [19], the correlation of Ruthven [10] based on a statistical thermodynamics model, and the Langmuir isotherm identified in this work. In Fig. 5, the oxygen isotherm data of Wright [20], the correlation of Ruthven [10], and the linear isotherm identified in this work are presented. The match between the nitrogen data and the model isotherm is excellent. The difference in the two sets of oxygen data in Fig. 5 is probably due to the relatively small amount of oxygen absorbed on zeolite 5A, which requires a very accurate gravimetric measurement. The model identified isotherm bisects both sets of given oxygen data.

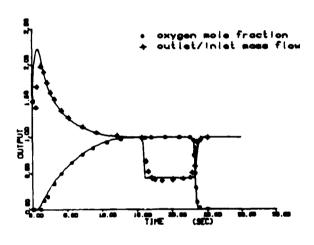
Comparison of Simulation and Experiment. Comparisons of the model, using the identified parameters, with the experimental data are shown in Fig. 3 and in Figs. 6 through 10. Shown in Fig. 3 are the data and simulation for bed pressure at $\frac{1}{2}$ atm for both oxygen and nitrogen with linear isotherms. As can be seen here for low pressure, a linear isotherm is adequate for predicting the time responses. The same values of K_A and K_B were then used for the 3 at-

SINGLE BED MODEL TECHNICAL DESCRIPTION APPENDIX A



INLET FLOW (GM/SEC)= 8.56
PRESSURE (MPA)= .3834
KA = .168
KB = .059
d = .21.1

Fig. 6 3 Atmosphere breakthrough data and simulation—linear isotherms.



INLET FLOW (GM/SEC)= 4.28 PRESSURE (MPA)= .0507 KA = .153 KB = .0506 b = 2.59

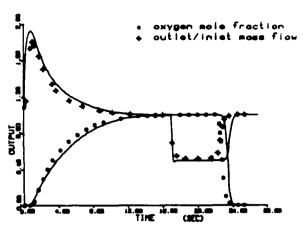
Fig. 7 $^{-1}2$ Atmosphere breakthrough data and simulation – nonlinear coupled isotherms

mosphere data, with the results of this simulation shown in Fig. 6. As can be seen, the match with experimental data is very poor. This indicates nonlinear isotherms are necessary.

Comparisons of the model using the nonlinear nitrogen isotherm with the experimental data are presented in Figs. 7 through 10., As seen in these figures, the data match has been greatly improved over the linear parameter model. Discrepancies between model and experiment can be attributed to experimental error and inaccurately assumed isotherms. The data indicate that a nonlinear oxygen isotherm may be appropriate, but this improvement is left for future work.

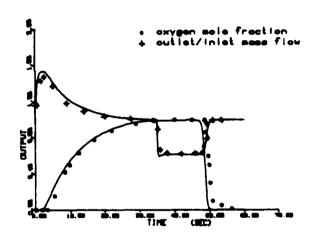
V Conclusions

A dynamic model of the O₂ - N₂ isobaric breakthrough



INLET FLOW (GM/SEC)= 8.48
PRESSURE (MPA)= .1884
KA = .153
KB = .8586
b = 2.59
d = 83.2

Fig. 8 1 Atmosphere breakthrough data and simulation – nonlinear coupled isotherms



INLET FLOW (GM/SEC) = 6.53
PRESSURE (MPA) = .2041
KA = .153
KB = .0506
b = 2.59
d = 27.5

Fig. 9 2 Atmosphere breakthrough data and simulation – nonlinear coupled isotherms

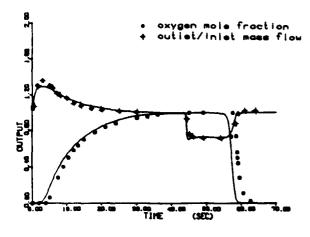
process in a molecular sieve bed was developed and validated experimentally. This model includes nonlinear and coupled isotherms. The proposed model accurately predicts the outlet mole fraction and outlet mass flow. In addition, isotherms for oxygen and nitrogen were identified which are in agreement with previously obtained isotherms.

This model is presently being used in an overall system model for a pressure swing adsorption process to generate oxygen from aircraft engine bleed air.

References

- 1 Breck, D. W., Zeolite Molecular Sieves, Wiley, 1974
- 2 Miller, R. L., et al., "Molecular Sieve Generation of Aviators Gregen.

APPENDIX A



INLET FLOW (GM/SEC)= 8.56 PRESSURE (MPA)= .3034 KA - .153 KB = 0500 = 2.59 ь 21.1

Fig. 10 3 Atmosphere breakthrough data and simulation - nonlinear coupled isotherms

Performance of a Prototype System Under Simulated Flights Conditions," Aviation Space and Environmental Medicine, July 1980, pp. 665-673.

- 1 Chan, Y. I., Hill, F. B., and Wong, Y. W., "Equilibrium Theory of a Pressure Swing Adsorption Process," Chem. Eng. Sci., Vol. 36, 1981, pp. 241-251
- 4 Knaebel, K. S., and Hill, F. B., "Equilibrium Theory of Oxygen Enrichment by Pressure Swing Adsorption," Presented at 1981 ASME WAM,
 - 5 Shendaiman, L. H., and Mitchell, J. E., "A Study of Heatless Ad-

sorption in the Model System CO2 ub He, I," Chem. Eng. Sci., Vol. 27, 1972 pp. 1449-1458.

- 6 Maurer, R. T., "Apparent Surface Diffusion Effects for Carbon Dioxide/Air and Carbon Dioxide/Nitrogen Mixtures with Pelleted Zeolite Beds." in Adsorption and Ion Exchange with Synthetic Zeolites, W. H. Flank, ed., 1980.
- 7 Barrer, R. M., Zeolites and Clay Minerals as Sorbents and Molecular Sieves, Acad. Press, 1978.
- 8 Bullow, M., Grossman, A., and Schirmer, W., "Zur Adsorption von binären Gasgemischen an festen Oberflächen: Möglichkeiten der Bestimmung von Gleichgewichtsparametern der Gemische auf der Grundlage der eni-sprechenden Einzelgasadsorptionsgleichgewichte," Zeit. Chemic, Vol. 12, 1972, pp. 161-169.
- 9 Penze, T., Bullow, M., and Schirmer, W., "Eine Methode zur Berechnung partieller Gemischadsorptionswerte aus den Daten der Einzeikomonentenadsorption," Z. Phys. Chem., Vol. 255, 1974, pp. 765-772.
- 10 Ruthven, D. M., "Sortion of Oxygen, Nitrogen, Carbon Monoxide, Methane, and Binary Mixtures of these Gases in SA Molecular Sieve," AlchE Jour., Vol. 22, 1976, pp. 753-759.

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- 11 Woltman, A. W., and Hartwig, W. H., "The Solution Theory Modeling of Gas Adsorption on Zeolites," in Adsorption and Ion Exchange with Synthetic Zeolites, W. H. Flank, ed., 1980.
- 12 Meyers, A. L., and Prausnitz, J. M., "Thermodynamics of Mixed Gas Adsorption," AIChE Jour., Vol. 11, 1965, pp. 121-127.

 13 Bering, B. P., and Serpinskii, V. V., "Substantiation of the Method of an
- Ideal Adsorption Solution for Calculating the Adsorption of Binary Vapor Mixtures from Individual Isotherms," Izv. Akad. SSSR, Ser. Khim. Plenum (Translation), 1972, pp. 158-160.
- 14 "Calculation of Adsorption Equilibrium of Binary Vapor Mixtures for the Case of Affined Isotherms," Izv. Akad, SSSR, Ser. Khim, Plenum, 1972, pp. 155-157.
- 15 "Determination of Activity Coefficients of Binary Adsorption Solutions," Izv. Akad. SSR, Ser Khim, 1972, pp. 152-154.
- 16 Hamming, R. W., Introduction to Applied Numerical Analysis, McGraw-Hill, 1971, pp. 215-237.

 17 Werlin, J. P., "Dynamic N₂/O₂ Separation Response of a Single Bed
- Molecular Sieve," Master's thesis, University of Texas at Austin, Dec. 1981.
- 18 Lightfoot, E. N., et al., in New Chemical Engineering Separation Techniques, Schoen, et al., ed., Interscience, 1962.
- 19 Peterson, D., "Influence of Preadsorbed Water on the Sorption of Nitrogen by Zeolites at Ambient Temperature," in Adsorption and Ion Ex-
- change with Synthetic Zeolites, W. H. Flank, ed., 1980.

 20 Wright, R. M., et al., "Development of Design Information for Molecular Sieve Type Regenerative CO₂ Removal Systems," NASA CR-2277. Prepared for NASA Langley, July 1973.

APPENDIX B

This article reprinted from the TRANSACTIONS OF THE ASME, JOURNAL OF DYNAMIC SYSTEMS, MEASUREMENT, AND CONTROL. The author is J.J. Beaman.

A Dynamic Model of a Pressure Swing Oxygen Generation System¹

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A full system model for oxygen generation from a pressure swing adsorption process is developed. The system under study has two zeolite packed molecular sieve beds which are cycled with high and low pressure in order to produce oxygen enriched air. These On-Board Oxygen Generation Systems (OBOGS) are to be used for the generation of aircraft crew breathing requirements. These units are much smaller in scale than present industrial systems and are called on to respond to variable demand. These special requirements have led to a reliable computer-based model to simulate the relatively fast transient performance of OBOGS. Besides the two zeolite beds, the system consists of a rotary valve used to cycle the beds, purge orifice, and breathing plenum. The computerized model predictions compare favorably with experiment. The model is presently being used in design and control synthesis studies.

1 Introduction

As mentioned in [1], a promising system for generating breathing oxygen for aircraft crew uses a synthetic zeolite material to physically adsorb nitrogen from engine bleed air. One such system is based on a "pressure swing adsorption cycle" in which nitrogen is adsorbed at high pressure and then desorbed at and exhausted to low pressure. The total process produces oxygen enrichment of air in a cyclical fashion. As a first step in the modeling of such systems, an experimentally verified model of a single bed system containing a zeolite molecular sieve bed was developed in [1]. The model developed in [1] provides predictions of the bed's transient outlet flowrate and mole fraction for a given inflow and mole fraction. The model was tested for pressures from 1/2 to 3 atmospheres and inlet mass flowrates of 3.21 gm/s to 8.56 gm/s.

Related work in this area can be found in Turnock and Kadlec [2] who study the sorption of nitrogen and methane with pressure swing adsorption. Wright et al. [3] develop a system model for a thermal swing adsorption cycle for CO₂ removal. Chan et al. [4] develop an equilibrium theory for pressure swing adsorption. Fernandez and Kenney [5] study a single column pressure swing separation of oxygen and nitrogen. Chihara and Suzuki [6] develop a nonisothermal pressure swing adsorption model. Carter [7] studies pressure swing adsorption for air drying.

Molecular sieve zeolite separation processes are in widespread use industrially, but OBOGS (acronym for On-Board Oxygen Generation System) units are much smaller in scale and are called on to respond to a much more variable demand as determined by the aircraft crew's breathing requirements. These special requirements have led to the need for a reliable computer-based simulation tool that would

model the relatively rapid transient response in OBOGS. This model will be useful for both design modification and control development.

2 Description of System

Shown in Fig. 1 is a schematic of a two bed OBOGS developed by Bendix Corporation for use in military aircraft. The object of the system model is to simulate the transient oxygen concentration delivered to the pilot given the pilot's (time varying) breathing mass flowrate. This is accomplished by mathematically modeling the rotary valve, purge orifice, the two zeolite beds, and the breathing plenum, and coupling these separate models into an overall system model.

The pressure swing adsorption cyclic operation of the beds can be described as:

- 1. The inlet of bed 1 is supplied air at supply pressure through the rotary valve, while bed 2 is being purged to exhaust pressure through the valve. During this stage breathing flow is obtained from bed 1.
- 2. As the rotary valve turns, the inlet of bed 1 is switched to exhaust pressure and the inlet of bed 2 is switched to supply pressure. During an initial transition stage, the pressure in bed $1,P_1$, is still higher than the pressure in bed $2,P_2$. In this case, there is flow out of both sides of bed 1 and into both sides of bed 2. Breathing flow will come from the higher pressure bed
- 3. As the pressure drops in bed 1 due to outflow and builds in bed 2 due to inflow, P_2 will become greater than P_1 . This is the identical case of (1) above, but with bed 2 the higher pressure side. The stages 1, 2, and 3 above describe 1/2 of a production cycle. The other half cycle is symmetric to the first half cycle.

The components of the system are the zeolite molecular sieve beds, rotary valve, bypass or purge valve between the two beds, and the breathing plenum. These components are modeled as discussed below.

¹This work supported by Brooks Air Force Base, Aerospace Medicine, Crew Technology

Contributed by the Dynamic Systems and Control Division for publication in the JOHANAL OF DYNAMIC SYSTEMS, MAASUREMENT, AND CONTROL. Manuscript received at ASME Headquarters, February 19, 1985.

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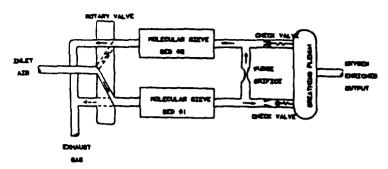
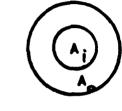


Fig. 1 Schematic of OBOGS



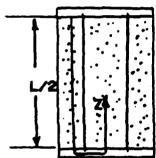


Fig. 2 Concentric zeolite bed

2.1 Zeolite Bed Model. Concentric beds were used in the experimental study and one is depicted in Fig. 2. As discussed in [1], if we assume one dimensional flow, negligible axial diffusion, diffusion into the zeolite described with a mass transfer coefficient, convective terms dominate temporal

changes in the free stream concentration, and isobaric beds (the pressure drop across the beds is small), the oxygen/nitrogen adsorption in the beds can be modeled as

$$\frac{\partial(uA)}{\partial z} = A\beta(\gamma/C - 1) \tag{1}$$

$$\frac{\partial (yA)}{\partial z} = A\beta(\gamma_A - y/u) \tag{2}$$

$$\frac{\partial n}{\partial t} = -d(\gamma - C) \tag{3}$$

$$\frac{\partial n_A}{\partial t} = -d(\gamma_A - C_A) \tag{4}$$

where

A = bed cross-sectional area

d = overall mass transfer coefficient

 $\beta = d(1 - \epsilon)/\epsilon$; $\epsilon = \text{void fraction of gas volume to total volume}$

 $C = C_A + C_B$ total gas phase concentration

 $C_A =$ oxygen gas phase concentration

 $\gamma = \gamma_A + \gamma_B$ total interfacial concentration

 $\gamma_A =$ oxygen interfacial concentration

u = convective velocity

 $n = n_A + n_B$ total adsorbed phase concentration

 $n_A =$ oxygen adsorbed phase concentration

 $y = uC_A$ convective oxygen flux

Due to the relatively fast heat transfer, it was observed that

Nomenclature -

A = bed cross-sectional area, m²

 $A_d = \text{orifice area, m}^2$

 $A_a = \text{inner annulus area, m}^2$ $A_a = \text{outer annulus area, m}^2$

 $C = \text{gas concentration, kgmole/m}^3$

 C_d = valve discharge coefficient

C_p = breathing plenum gas concentration, kgmole/m³

d = total mass transfer coefficient, 1/s

k = ratio of specific heats

L = bed length, m

M = average molecular weight of gas in plenum, kg/kgmole

 $n = \text{adsorbed phase concentration, kgmole/m}^3$ $P_0, P_1, P_2, P_3 = \text{supply, Bed 1, Bed 2, exhaust pressures,}$

 (N/m^2) $Q = \text{volume flowrate, } m^3/s$

t = time, s

u = convective velocity, m/s

If = mass flowrate, kg/s

 $\nu = \text{molecular flux of oxygen, kgmole/m}^2-s$

z =bed distance, m

 $\beta = d(1 - \epsilon)/\epsilon (1/s)$

 $\gamma = interfacial concentration, kgmole/m³$

 $\Delta t = \text{time step, s}$ $\Delta z = \text{space lump, m}$

c = space rump, m

 ϵ = void fraction

A = denotes property of oxygen
 B = denotes property of nitrogen

BR = denotes breathing property

i = space index

j = time index

L = denotes property at z = L

P =denotes purge orifice property

0 = denotes property at z = 0

1 = denotes bed 1 property

2 = denotes bed 2 property

Superscripts

= denotes property just before z = L/2

+ = denotes property just after z = L/2

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the experimental beds remained at essentially constant temperature during a cycle even though the adsorption process is not intrinsically isothermal. A more detailed study of the thermal properties of the system is the subject of engoing research. In the isothermal case, γ_4 and γ_B , the interfacial concentrations of oxygen and nitrogen, respectively, are related to n_4 and n_B , the adsorbed oxygen and nitrogen concentrations, by the equilibrium adsorption isotherms [8]. As discussed in [1], these isotherms can be expressed as

$$\gamma_A = \gamma_A (n_A, n_B) \tag{5a}$$

$$\gamma_B = \gamma_B (n_A, n_B) \tag{5b}$$

for nonlinear coupled isotherms. The pure component isotherms were identified as

$$\gamma_4^* = K_4 n_4^* \tag{6a}$$

$$\gamma_B^* = K_B n_B^* (1 - n_B^*/b) \tag{6b}$$

with $K_A = 0.153$ kgmole gas phase/kgmole adsorbed phase, $A_B = 0.0506$ kgmole gas phase/kgmole adsorbed phase, and $b \approx 2.59$ kgmole adsorbed phase. The coupling is predicted by using an ideal adsorption method [1].

The diffusion coefficient d was empirically chosen as 200 1 s in order to best match the data. This value is higher than the values identified in [1]. This is probably due to the high velocities in the beds during exhaust and initial intake of air.

2.2 Valve Model. The orifices of the rotary valve and the purge orifice are modeled as standard compressible flow pneumatic orifices as described in [9]. The basic equation for the flow of a gas through an orifice is of the following form

$$W = C_d A_d f\left(P_u, T_u, \frac{P_d}{P_u}\right) \tag{7}$$

where

W = mass flowrate

A, is orifice area

 C_f is a dimensionless discharge coefficient

 T_u is upstream stagnation temperature

 P_{ω} is upstream stagnation pressure

 P_J is downstream pressure

f is a function of P_u , T_u , and P_d/P_u .

For pressure ratios P_d/P_u greater than the choked flow ratio and less than 1, the function f can be expressed as

$$f = C \cdot \frac{P_u}{T_u^{1/2}} (P_d/P_u)^{1/k} (1 - (P_d/P_u)^{(k-1)/k})^{1/2};$$
 (8)

for pressure ratios less than the choked flow ratio.

$$f = C_2 \frac{P_u}{T!^2},$$

where C_1 and C_2 are functions of k, the ratio of specific heats, and the gas constant. These parameters are taken as constants for this study. For the rotary valve the orifice areas are cyclic functions of the rotation angle of the valve. The discharge coefficients were chosen between values of 0.6 and 1. in order to best match the measured experimental flow.

2.3 The Breathing Plenum. The breathing plenum is modeled as an ideal isobaric two stage mixing chamber. The transfer function between inlet mole fraction and outlet mole fraction can be expressed as:

$$\frac{x_a}{x_c} = \frac{1}{(75+1)^2}$$
 (9)

where

$$\tau = \frac{C_p V_{BR} \dot{M}}{W_{BB}}$$



Fig. 3 Conceptual bond graph of system

 W_{BR} is breathing mass flowrate, C_p is the plenum molecular concentration, V_{BR} is approximately half of the plenum volume, \bar{M} is an average molecular weight. In deriving these equations the following assumptions have been made:

- 1. The plenum remains at approximately constant pressure and therefore constant C_p . This is not strictly true as breathing oxygen is being switched between the beds, but for our prototype system this assumption is reasonable.
- 2. The molecular flux into and out of the plenum can be found approximately from the mass flowrate and an average molecular weight. This assumption is quite reasonable for oxygen and nitrogen since their individual molecular weights do not differ substantially.
- 3. The mixing volumes for the two idealized chambers are equal and can be found from half the plenum volume plus a correction for mixing in the piping.

3 Solution Technique

- 3.1 Causal Considerations. The overall system model can be depicted by the word bond graph [10] shown in Fig. 3. Besides the physical interconnections, the graph also indicates the important causal relationships in the system model. The causality shown in the graph can be read as:
- 1. Pressures P_0 and P_1 are supplied as inputs to the rotary valve while the valve determines the flowrate Q_{01} from these two pressures.

William Princes

- 2. P_1 is determined from bed 1 which has inputs the flow Q_{01} and the flow Q_{L1} .
- 3. Pressures P_1 and P_2 are the inputs to the purge valve and the valve supplies the flowrates Q_{P1} and Q_{P2} to the system. Q_{BP1} and Q_{BP2} are determined by the breathing flow and bed pressures in the check valve/plenum system as described in the next section.
- 4. P_2 is determined by bed 2 which has inputs the flows Q_{02} and Q_{L2} .
- 5. Q_{L2} is determined by the rotary valve with inputs to the valve of P_2 and P_3 .

In addition to these causal statements, the spatial molecular concentration of the two gases in the beds is determined in the direction of the flowrate i.e., the concentration must be determined from upstream conditions. This will be discussed more thoroughly in the next section.

The causality described above and shown on the graph is integral causality for the beds if they are treated as capacitive fields [11] or mass storage devices. Using other causal arrangements is possible but will result in complicated and computationally time consuming iteration.

- 3.2 Solution Algorithm. Given input supply pressures P_0 and P_1 , breathing flow W_{BR} , and initial adsorbed phase concentrations n(z,0) and $n_A(z,0)$ in the two beds, the system equations can be solved as a boundary value problem in space and a marching problem in time. The solution technique consists of:
- (i) Given n(z,0) and $n_A(z,0)$, the interfacial gas phase concentrations $\gamma(z,0)$ and $\gamma_A(z,0)$ can be calculated from the equilibrium isotherms (5). Equation (1) can then be integrated along the entire length of each bed as

$$(Q_1 - Q_0 + A_1 \beta L/2 + A_0 \beta L/2)C$$

$$=A_0\beta\int_0^{L/2}\gamma dz + A_1\beta\int_{L/2}^L\gamma dz. \tag{10}$$



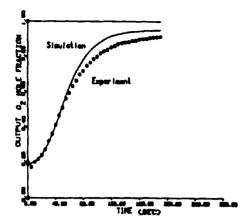


Fig. 4 Oxygen mole fraction varsus time (10 (IllUmin) breathing flow)

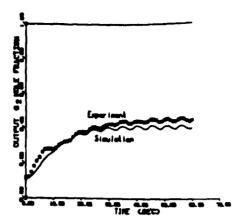


Fig. 6 Oxygen mole fraction versus time (50 (lit/min) breathing flow)

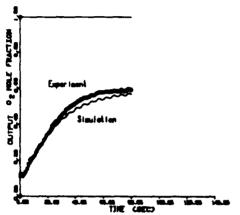


Fig. 5 Oxygen mole fraction versus time (30 (lit/min) breathing flow)

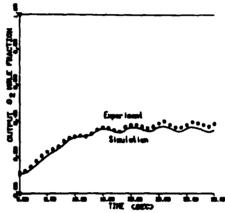


Fig. 7 Oxygen mole fraction versus time (70 (lit/min) breathing flow)

For bed 1, $Q_0 = Q_{01}(P_0, P_1)$ where P_1 is proportional to C_{11} the gas phase concentration in bed 1, for isothermal beds and assuming a perfect gas, and $Q_1 = Q_1(P_1, P_2) - Q_{BR1}$ where P_2 is proportional to C_2 , the gas concentration in bed 2 and Q_{BR} accounts for the volume flow out of bed I due to crew breathing. For product flow out of bed I, Quei is proportional to Was for an isobaric and isothermal phenum. For bed 2, $Q_c = Q_{02}(P_1, P_2) - Q_{BR2}$ and $Q_L = Q_{L2}(P_2, P_3)$ where Q_{BR} , accounts for breathing flow out of bed 2. Note that either Q_{BR1} or Q_{BR2} will be zero depending on which bed had the lower pressure. Equation (10) written out for both beds represents two coupled nonlinear algebraic equations for the two unknowns C_1 and C_2 . These equations can be solved by an iteration procedure consisting of (1) assuming concentration C_2 and solving the resultant single nonlinear algebraic equation for C_1 using a modified false position algorithm. (2) using this value of C_1 to calculate a new value of $C_{i,j}$ (3) iterate until the solution procedure converges. Convergence of this procedure requires on the order of three to four iterations. The integrations involved in (10) are performed numerically by a trapezoidal rule, and are only done once per time step. This iteration procedure amounts to a solution of the boundary value problem in the system.

(ii) Spatially integrate equation (1) numerically as

$$u = u_{-1} + \beta(\gamma_1 + \gamma_{-1})/2C - \beta\Delta z, \tag{11}$$

where t is the index for the space dimension, $z = t\Delta z$, and Δz is the space lump size. This integration is done in two steps from 0 to t = 2 and then from L/2 to L with condition $u(z^+) = A(z^-)u(z^-)/A(z^+)$ at z = L/2 to account for the area

change in the outer annulus and inner cylinder of the bed. Note that the starting velocity $u_0 = Q_0/A(0)$ is known from step (i).

(iii) Using the velocity distribution calculated in (ii), equation (2) is a first order linear ordinary differential equation in space for fixed time. Approximate the velocity u and the interfacial concentration γ_A over a space lump as

$$u=r_0+r_1z$$

$$(i-1)\Delta z < z < i\Delta z$$

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where

$$r_0 = u_{i-1}, r_1 = (u_i - u_{i-1})/\Delta z$$

$$r_2 = \gamma_{A_1-1}, r_3 = (\gamma A_1 - \gamma_{A_1-1})/\Delta z$$

Using these approximations, equation (2) can be integrated in closed form as shown in [1]. The direction of the integration will always be in the direction of the velocity flow. Which, as described in Section 2, might be (1) flow in one side of the bed and out the other, (2) flow into both sides of the bed, or (3) flow out of both sides of the bed. In cases (2) and (3), there will be at least one point in the bed that has zero flow. As the velocity in the bed is found as described in (ii) above, the point of zero velocity is stored. For case (1), the integration will proceed in the direction of flow, for case (2) the integration will proceed from the sides of the bed to the point of zero flow; and for case (3) the integration will start at the point of zero velocity and proceed out the two sides. Although physically possible, multiple points of zero flow will not occur.

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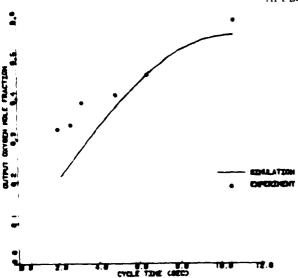


Fig. 8. Oxygen mole fraction versus cycle time (30 (lit/min) breathing flow).

in normal operation of OBOGS and this case is not included at present in the computerized solution.

The ability to solve equation (2) in closed form is an important numerical result. The oxygen profiles in the beds can be very steep, and if a completely numerical procedure is applied to (2), it will require small increments for Δz in order to be stable and accurate. This will result in very long computation times even on a high speed computer.

(n) Equations (3) and (4) are then integrated in time via a simple Euler method as

$$n_{i,j} = n_{i,j-1} - d(\gamma_i - C) \Delta t \tag{12}$$

$$n_{A_{i,j}} = n_{A_{i,j-1}} - d(\gamma_{A_i} - C_{A_i})\Delta t$$
 (13)

where j is the time index, $t = j\Delta t$, and Δt is the time step.

(c) The input oxygen mole fraction x_j to the breathing plenum is found from bed 1 at z=L or bed 2 at z=0 depending on which bed has higher pressure. The mole fraction from the higher pressure bed is computed as C_A/C . This input mole fraction is used with equation (9) and the breathing flow W_{BR} to solve for the output mole fraction, x_0 . Equation (9) is solved as two coupled ordinary differential equations via a simple Euler method.

(vi) March in time by proceeding to step (i) for the new time level.

The time step and space lumps were chosen by performing numerical experiments to insure accuracy and stability of the solution procedure. The model simulation time is approximately twice that of real time when computed on a CDC Cyber 750-150.

4 Comparison of Model With Experiment

The prototype experimental system is depicted in Figs. 1 and 2. The concentric molecular sieve beds have an inner area of 36.5 cm² and an effective outer area of 136. cm². The effective flow length of a bed is 39.4 cm. The beds are loaded with 1 inde 5.4 sorbent with zeolute particles of approximately 340 to 833 microns in diameter. The void fraction of the beds is r = 0.3. All experiments were run at room temperature of 25° C. No noticeable variation in temperature was observed during the system operation which implies the isothermal assumption is valid for the system under study. The fully open supply pressure valve has an effective diameter of .78 cm. and a discharge coefficient of 0.8. The fully open exhaust pressure valve has an effective diameter of 1.12 cm and a discharge

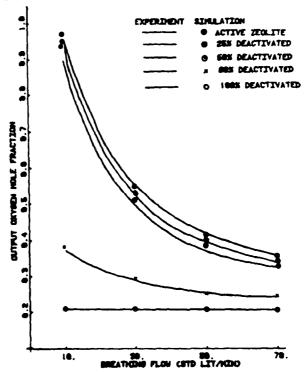


Fig. 9 Effect of reducing active sleve volume

coefficient of 0.8. The purge orifice has a diameter of 0.191 cm and a discharge coefficient of 0.75. The breathing plenum volume is .0018 m³ an additional volume of .0006 m³ is added to this quantity to account for mixing in the connections.

In this paper, comparison with of the model is made with three basic types of experiments: (1) transient startup, (2) cycle time variation, and (3) change of active volume of zeolite in the beds.

4.1 Transient Startup of OBOGS. Shown in Figs. 4 through 7 are the results from the experimental data and the simulation model for the OBOGS at initially atmospheric pressure, the rotary valves turned on at time = 0, a supply pressure of 0.31 Mpa, an exhaust pressure of .10 Mpa, and cycle time of 10.7 seconds. These data are shown for four different constant breathing flowrates of 0.214, 0.642, 1.07, and 1.50 gm/min (10, 30, 50, and 70 standard liters/min). As can be seen in these figures, the model does an excellent job of matching the transient experimental data for the breathing oxygen mole fraction over the range of breathing flowrates. The increased ripple in the data and simulation for increased flow is due to less time for complete mixing in the plenum as the breathing flow goes up. The model describes this effect very well.

These plots indicate a method for controlling the outlet mole fraction. This is desirable since both high and low oxygen mole fraction can cause physiological problems [12] to the aircraft crew. By dumping product gas to artificially increase the breathing flow, the outlet mole fraction can be lowered if too high. If the mole fraction is too low, product gas dumping can be decreased.

4.2 Effect of Cycle Time Variation. In Fig. 8, the experimental and simulation results for a variation in cycle time are shown. As can be seen, the model does a good job of describing changes in oxygen output with cycle time variation. The results in this figure also demonstrate a method for controlling the outlet oxygen. By simply varying the cycle time of the process, control of oxygen output is possible. The cycle

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time variation can be easily implemented by using a variable speed drive for the rotary valve. This control scheme has the advantage of not using any extra product gas which might be destrable for the contaminant filtering properties of the beds. The study of this control scheme is the subject of ongoing research and these results will be presented at a later date.

4.3 Bed Volume Variation. Figure 9 shows the effect of changing the volume of the beds, this was done experimentally [13] by filling the beds with fractional amounts of molecular sieve which had been deactivated with water. From simulation results, it was found that only bed volume and not shape is of importance for oxygen output. This might be an important design result, for it implies that the shapes of the beds are not critical. This means that the beds could be shaped for space considerations in the aircraft.

Another interesting result, is the relative insensitivity of the beds to volume changes until a critical volume is reached (around 75 percent for these beds). This also has important design implications. It means that sizing the system for increased flow is not just a simple increase in bed size, but a combination of volume changes and other design parameters. In particular, the valve areas and discharge coefficients are important design values.

In addition, these results indicate a potential problem in operation. If the beds were to become contaminated with water, which has the effect of decreasing the effective volume of the beds, there would be no outward indication of performance degradation until a critical volume were reached, and at that point the beds would rapidly lose performance with any further contamination.

5 Conclusions

A relatively simple model of a pressure swing oxygen generation system was developed in this study. The computerized model includes the zeolite beds, valves, and plenum breathing chamber in the system. The model predictions compare favorably with detailed experimental data. Ongoing

work in this area include the prediction of temperature and contaminant effects and control of output breathing oxygen.

Acknowledgments

The author gratefully acknowledges the experimental support of Dr. K. Ikels and Mr. C. Theis at Brooks. In addition, the author wishes to thank Mr. J. Werlin and Dr. S. Wang, who were graduate research assistants, for collecting much of the experimental data.

References

- Beaman, J. J., Healey, A. J., and Werlin, J., "A Dynamic Model of a Molecular Sieve Bed with Nonlinear and Coupled Isotherms," ASME JOURNAL OF DYNAMIC SYSTEMS, MEASUREMENT AND CONTROL, Vol. 105, No. 4, Dec. 1983, pp. 265-271.
- 2 Turnock, P. H., and Kadlec, R. H., "Separation of Nitrogen and Methane Via Periodic Adsorption," *AICHE Journal*, Vol. 17, No. 2, 1971, pp. 335-342.
- 3 Wright, R. M., et al., "Development of Design Information for Molecular Sieve Type Regenerative CO₂ Removal Systems," NASA CR-2277, prepared for NASA Langley, July 1973.
- 4 Chan, Y. I., Hill, F. B., and Wong, Y. W., "Equilibrium Theory of a Pressure Swing Adsorption Process," Chem. Eng. Sci., Vol. 36, 1981, pp 243-251.
- 5 Fernandez, G. F., and Kenney, C. N., "Modelling of the Pressure Swing Separation Process," Chem. Eng. Sci., Vol. 38, No. 6, 1983, pp. 827-834.
 6 Chihara, K., and Suzuki, M., "Simulation of Nonisothermal Pressure
- Swing Adsorption," Jour. Chem. Eng. Japan, Vol. 16, No. 1, 1983, pp. 53-60.
 7 Carter, J. W., and Wyszynski, M. L., "The Pressure Swing Adsorption Drying of Compressed Air," Chem. Eng. Sci., Vol. 38, No. 7, 1983, pp. 1081.
 - 8 Breck, D. W., Zeolite Molecular Sieves, Wiley, 1974.
- 9 Blackburn, J. F., Resthof, G., and Shearer, J. L., Fluid Power Control, MIT Press, 1960.
- 10 Paymer, H. M., Analysis and Design of Engineering Systems, MIT Press,
- 11 Karnopp, D. C., and Rosenberg, R. C., System Dynamics: A Unified Approach, Wiley, 1974.
- 12 Ernsting, J., Internal memorandum and private communication, RAF In-
- strute of Aviation Medicine, Farnborough, United Kingdom, 1982.

 13 Tkels, K., and Theis, C., "The Effects of Moisture on Molecular Sieve Oxygen Concentrators," Aviation Space and Environmental Medicine, to appear.

COMPUTER MODEL FLOWCHART

1. Governing Equations

(1)
$$\frac{\partial(UC)}{\partial Z} = \beta(\gamma - C)$$
 $\beta \equiv \frac{d - (1 - \epsilon)}{\epsilon}$

(2)
$$\frac{\partial (UC_A)}{\partial Z} = \beta(\gamma_A - C_A)$$
 $\eta \equiv \eta_A + \eta_B$

(3)
$$\frac{\partial \eta_A}{\partial t} = -d(\gamma_A - C_A)$$
 $\gamma \equiv \gamma_A + \gamma_B$

(4)
$$\frac{\partial \eta}{\partial t}$$
 $-d(\gamma - C)$

$$(5) \quad \frac{\partial \mathbf{C}}{\partial \mathbf{Z}} = \mathbf{0}$$

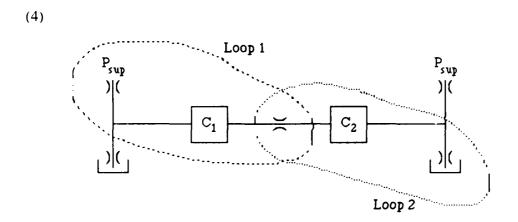
(6) W
$$C_d \Lambda \sqrt{\frac{2k}{R(k-1)}} \frac{P_u}{\sqrt{T}} \left[\frac{P_d}{P_u} \right]^{1/k} \sqrt{1 - \left[\frac{P_d}{P_u} \right]^{\frac{k-1}{k}}}$$

II Procedures

- (1) Initial guess P(1) and P(2)
- (2) Use adsorption isotherms to calculate γ_A and γ_B
- (3) Integrate equation (1) to get C:

$$(U_{\beta} A_{\alpha} + U_{\alpha} A_{\beta}) C = \beta \int (\gamma - C) dz$$

$$C = \frac{1}{(U_1|A_0 + U_0|A_1)}\beta \int (\gamma + C) \ dz$$



In Loop 1 and Loop 2, use False Position Method to find C_1 and C_2 so that these

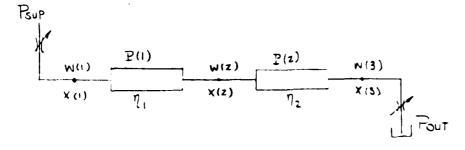
COMPUTER MODEL FLOW CHART AND SAMPLE INPUT AND OUTPUT APPENDIX C

values will be consistent with the supply pressure and exhaust pressure at each time step.

- (5) Substitute P(1) and P(2) (calculated in step 4) into the valve eq. can get w(1), w(2), and w(3)
- (6) Since Ui = W / (Ci*Ai*Mi), where Mi is the molecular weight, U_{01} , U_{11} , and U_{02} and U_{12} can be calculated.
- (7) Assume velocity is linear along the bed length. By integrating eq. (2), CA(Z) can be obtained.
- (8) Use eq. (3) and eq. (4) to get new values for η_A , η_B and η for each time step.
- (9) T = T + dT.
 Get a new valve area and repeat the procedure again.

III. Flowchart of the computer program

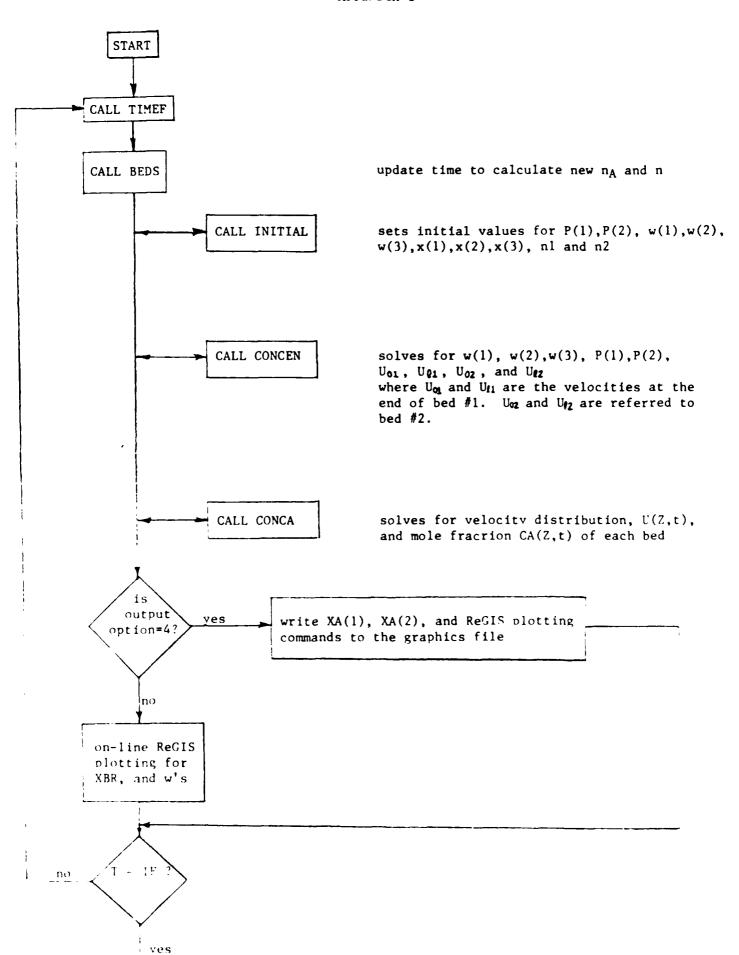
(1) Sign Convention

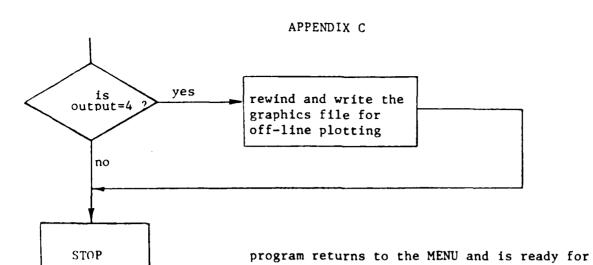


(2) Flowchart

Please see attached.

COMPUTER MODEL FLOW CHART AND SAMPLE INPUT AND OUTPUT APPENDIX C





the next run

STOP

The inputs to the program are, as in the menu:

SYSTEM PARAMETERS		
l. supply pressure	=	40.00 PSIA
2. outlet pressure	=	15.00 PSIA
3. final observation time	=	30.00 SEC.
4. cycle time	=	10.70 SEC.
5. breathing flowrate	=	10.00 STD LIT/MIN
GEOMETRIC PARAMETERS		
6. by-pass valve diameter	=	0.0750 IN.
7. supply valve diameter	=	0.3062 IN.
8. outlet valve diameter	=	0.4387 IN.
9. bed length	=	15.50 IN.
10. outer bed diameter	-	5.7300 IN.
11. inner bed diameter	=	2.1800 IN.
BED PARAMETERS		
12. KA	=	0.1530 KGMOLE O2 ABSORBED/KGMOLE GAS
13. KB	=	0.0490 KGMOLE N2 ABSORBED/KGMOLE GAS
14. B	=	2.6500
15. diffusion coefficient	=	200.00 1/SEC
16. void fraction	=	0.37

The user has the following options after he has reviewed the menu:

- 1. change parameters
- 2. insert a step change in the breathing flowrate, or
- 3. exit the program

First, changing the parameters is done by entering the corresponding parameter number appeared in the menu. The program then prompts, for example:

ENTER THE PARAMETER AND PRESS "RETURN"

CURRENT VALUE

SUPPLY PRESSURE (PSIA)

40.00

Once the new value has been assigned, the program goes back to the menu with modified parameter(s) and waits for the next input.

If, on the other hand, the user wishes to insert a step change in breathing flowrate, he enters the number for that option (which is "59" in this program). The program will ask for the starting and ending time, as well as the value for the step change and again goes back to the menu.

After the completion of modifications on the parameters, the user enters "0" to run simulation. The program responds with:

122224 E

TELLEGAL RESERVED RECESSES PLACESSES PARTIES

APPENDIX C

THE FOLLOWING PLOTS ARE AVAILABLE TO YOU ON THE TERMINAL

- 1 MOLE FRACTION OF OXYGEN VS.TIME (DATA FILE=MOLEFRAC.DAT)
- 2 INLET MASS FLOWRATE VS. TIME (DATA FILE=FLOWRATE.DAT)
- 3 OUTLET MASS FLOWRATE VS. TIME (DATA FILE=FLOWRATE DAT)
- 4 A DYNAMIC SIMULATION FOR ONE CYCLE (OXYGEN VS. TIME, DATA FILE=PROFILE.DAT)

PRESS THE CORRESPONDING # TO HAVE THE OUTPUT SHOWN AS THE SIMULATION TRANSPIRES

The user chooses the desired output, and the program gives the following:

- 1 ENTER "0" TO RUN SIMULATION.
- 2 ENTER "19" TO CHANGE THE PARAMETERS
- 3 ENTER "99" TO EXIT PROGRAM.

EMIER THE CORRESPONDING # AND PRESS "RETURN"

and waits for the decision from the user. That is, it is still possible for the user to go back to the menu before he runs the simulation.

Tradly, the program asks for a last item:

10 200 WISH 10 DOUBLE THE NUMBER OF SPACE LUMPS IN
18 (IMULATION FOR BETTER ACCURACY 2 (CURRENT NUMBER USED
15 50) (Y/N)

1 is a six due to the fact that when using one hundred lumps the sorth ation is time consuming for the low breathing flowrate cases. It is default is thus set to be 50 for faster simulation.

```
CURRENT SYSTEM PARAMETERS ARE:
 1 SUPPLY PRESSURE= 40.00 PSIA
OUTLET PRESSURE= 15.00 PSIA
                           15.00 PSIA
   FINAL OBSERVATION TIME= 30.00 SEC
 FOYCLE TIME= 10.70 SEC BREATHING FLOWRATE= 10.00 STD LIT/MIN
 4 CYCLE TIME=
 CURRENT GEOMETRIC PARAMETERS ARE:
 3Y-PASS VALVE DIAMETER = 0.0750 IN
   SUPPLY VALVE DIAMETER = 0.3062 IN
3 OUTLET VALVE DIAMETER = 0.4387 IN
 ∃ BED LENGTH =
                         15.5000 IN
10 OUTER BED DIAMETER = 5.7300 IN
11 INNER BED DIAMETER = 2.1800 IN
 THE CURRENT BED PARAMETERS ARE:
1 ' KA :
                            0.1530 KGMOLES 02 ABSORBED/KGMOLE GAS
13 KB =
                            0.0490 KGMOLES N2 ABSORBED/KGMOLE GAS
14 B =
                            2.6500
15 DIFFUSION COEFFICIENT = 200.00 1/SEC
16 VOID FRACTION =
                            0.370
ENTER CORRESPONDING PARAMETER # TO CHANGE PARAMETER
ENTER """ TO SIGNIFY PARAMETERS ARE CORRECT -- RUN SIMULATION
ENTER "59" TO INSERT A STEP CHANGE IN BREATHING FLOWRATE
ENTER "99" TO EXIT PROGRAM
THER THE PARAMETER AND PRESS "RETURN"
                                    CURRENT VALUE
 45 % PRESSURE (PSIA)
                                     40.0000
```

```
(URRENT SYSTEM PARAMETERS ARE
1 SUPPLY PRESSURE= 30.00 PSIA
                         15.00 PSIA
   OUTLET PRESSURE=
  FINAL OBSERVATION TIME= 30.00 SEC
 1 CYCLE TIME=
                          10 70 SEC
  BREATHING FLOWRATE=
                      10.00 STD LIT/MIN
 CURRENT GEOMETRIC PARAMETERS ARE:
BY-PASS VALVE DIAMETER = 0.0750 IN
   SUPPLY VALVE DIAMETER = 0 3062 IN
   OUTLET VALVE DIAMETER = 0.4387 IN
   BED LENGTH =
                        15.5000 IN
10 OUTER BED DIAMETER =
                         5.7300 IN
1: INNER BED DIAMETER =
                          2.1800 IN
 THE CURRENT BED PARAMETERS ARE:
11 KA =
                          0.1530 KGMOLES 02 ABSORBED/KGMOLE GAS
1 · KB =
                          0.0490 KGMOLES N2 ABSORBED/KGMOLE GAS
14 B =
                          2.6500
15 DIFFUSION COEFFICIENT =
                          200.00 1/SEC
16 VOID FRACTION =
                           0.370
ETITER CORRESPONDING PARAMETER # TO CHANGE PARAMETER
ENTER "9" TO SIGNIFY PARAMETERS ARE CORRECT -- RUN SIMULATION
ENTER "59" TO INSERT A STEP CHANGE IN BREATHING FLOWRATE
ENTER "99" TO EXIT PROGRAM
59
FRIER THE TIME AT WHICH THE STEP CHANGE
IF BREATHING FLOWRATE IS TO OCCUR
THE FINAL OBSERVATION TIME IS CURRENTLY 30.00 SECONDS
1.1
SHER THE TIME AT WHICH THE STEP CHANGE
I BREATHING FLOWRATE IS TO END
```

THE STREET OBSERVATION TIME IS CURRENTLY 30.00 SECONDS

*..****************

ENTER THE NEW BREATHING FLOWRATE OCCURING AS A STEP CHANGE FROM T= 10.00 TO T= 20.00 SECONDS

THE CURRENT BREATHING FLOWRATE IS 10.00 STD LIT/MIN

30

THE BREATHING FLOWRATE OF 30.00 STD LIT/MIN UTLL BE INPUT AS A STEP AT T= 10.00 SECONDS AND WILL END AT T= 20.00 SECONDS


```
CURRENT SYSTEM PARAMETERS ARE:
1 SUPPLY PRESSURE=
                     30.00 PSIA
 2 OUTLET PRESSURE=
                           15.00 PSIA
  FINAL OBSERVATION TIME= 30.00 SEC
                            10.70 SEC
   CYCLE TIME=
                       10.00 STD LIT/MIN
   BREATHING FLOWRATE=
 (URRENT GEOMETRIC PARAMETERS ARE:
6 BY-PASS VALVE DIAMETER = 0.0750 IN
7 SUPPLY VALVE DIAMETER = 0.3062 IN
8 OUTLET VALVE DIAMETER = 0.4387 IN
9 BED LENGTH =
                          15.5000 IN
10 OUTER BED DIAMETER =
                           5.7300 IN
                           2.1800 IN
   INNER BED DIAMETER =
  THE CURRENT BED PARAMETERS ARE:
12 KA =
                            0.1530 KGMOLES 02 ABSORBED/KGMOLE GAS
                            0.0490 KGMOLES N2 ABSORBED/KGMOLE GAS
13 KB =
                            2.6500
11 8 -
   DIFFUSION COEFFICIENT =
                            200.00 1/SEC
                             0.370
: "ID FRACTION =
ELE M COMPESPONDING PARAMETER # TO CHANGE PARAMETER
1 1 10 100 TO SIGNIFY PARAMETERS ARE CORRECT--RUN SIMULATION
Fig. 1959 TO INSERT A STEP CHANGE IN BREATHING FLOWRATE
   1 1991 O EXIT PROGRAM
```

THE FOLLOWING PLOTS ARE AVAILABLE TO YOU ON THE TERMINAL

- MOLE FRACTION OF OXYGEN VS.TIME (DATA FILE=MOLEFRAC.DAT)
- 2 INLET MASS FLOWRATE VS. TIME (DATA FILE=FLOWRATE.DAT)
- 3 OUTLET MASS FLOWRATE VS. TIME (DATA FILE=FLOWRATE.DAT)
- 4 A DYNAMIC SIMULATION FOR ONE CYCLE (OXYGEN VS. TIME, DATA FILE=PROFILE.DAT)

PRESS THE CORRESPONDING # TO HAVE THE OUTPUT SHOWN AS THE SIMULATION TRANSPIRES

1

- 1. ENTER "O" TO RUN SIMULATION.
- 2. ENTER "19" TO CHANGE THE PARAMETERS
- 3 ENTER "99" TO EXIT PROGRAM.

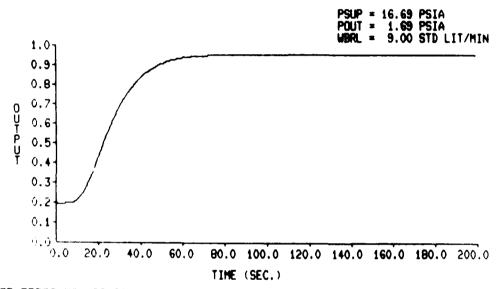
ENTER THE CORRESPONDING # AND PRESS "RETURN"

0

DO YOU WISH TO DOUBLE THE NUMBER OF SPACE LUMPS IN THE SIMULATION FOR BETTER ACCURACY ? (CURRENT NUMBER USED IN 50) (Y/N)

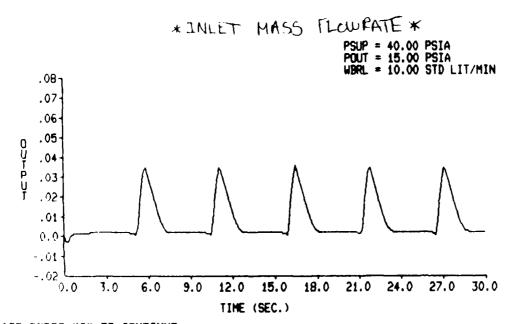
SAMPLE OUTPUTS

* OXYGEN MOLE FRACTION *



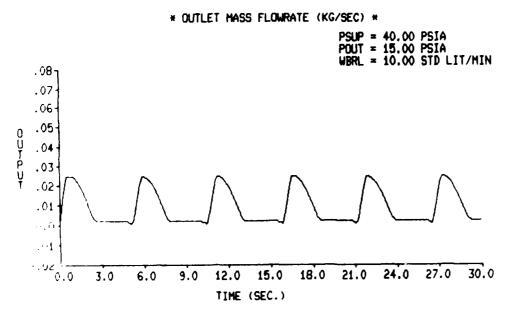
PLEASE PRESS "C" TO CONTINUE :

SAMPLE OUTPUTS

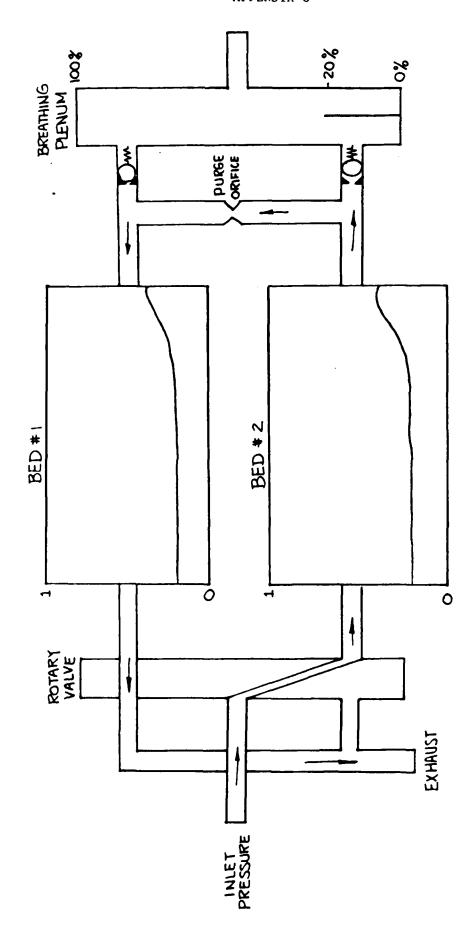


PLEASE PRESS "C" TO CONTINUE :

SAMPLE OUTPUTS



PLEASE PRESS "C" TO CONTINUE :



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